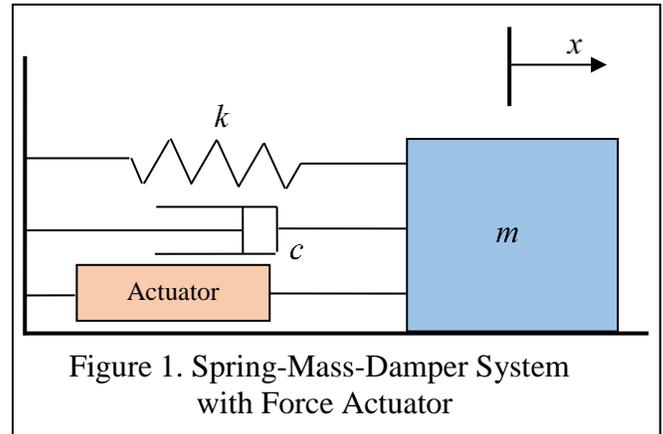


# ME 3600 Control Systems

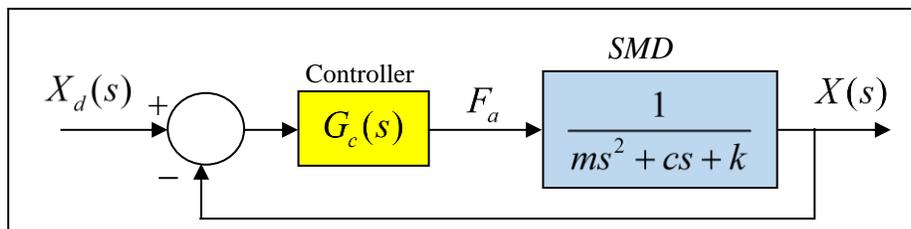
## PID Control of a Spring-Mass-Damper (SMD) Position

Fig. 1 shows a *spring-mass-damper* system with a *force actuator* for *position control*. The spring has stiffness  $k$ , the damper has coefficient  $c$ , the block has mass  $m$ , and the position of the mass is measured by the variable  $x$ . As discussed in earlier notes, the *transfer function* of the SMD with the actuating force  $F_a$  as input and the position  $x$  as output is



$$\boxed{\frac{X}{F_a}(s) = \frac{1}{ms^2 + cs + k}} \quad (1)$$

Assuming *ideal actuator* and *sensor* responses, the closed-loop position control of the SMD can be described using the following block diagram. Here,  $X_d$  represents the *desired position*, and  $G_c(s)$  represents the *transfer function* of the controller.



In the following analyses, the SMD parameters are assumed to be:  $m=1$  slug,  $c=8.8$  (lb-s/ft), and  $k=40$  (lb/ft). This represents an *under-damped, second-order* plant with

$$\omega_n = \sqrt{40} = 6.325 \text{ (rad/s)} \approx 1 \text{ (Hz)} \dots \text{ natural frequency}$$

$$\zeta = \frac{8.8}{2\sqrt{40}} = 0.696 \approx 0.7 \dots \text{ damping ratio}$$

### Proportional Control

If simple *proportional control* is used, then  $\boxed{G_c(s) = K}$ . In this case, the loop transfer function and closed-loop transfer functions are

$$GH(s) = \frac{K}{s^2 + 8.8s + 40}$$

$$\frac{X}{X_d}(s) = \frac{K}{s^2 + 8.8s + (40 + K)} \quad (2)$$

This is a *type-zero* system and hence will have a *finite steady-state error* for a step input. Using the *final-value theorem* and the *closed-loop transfer function*,  $x_{ss}$  the final value of  $x(t)$  to a unit step command is

$$x_{ss} = \lim_{s \rightarrow 0} \left( s \cdot \frac{1}{s} \cdot \frac{K}{s^2 + 8.8s + (40 + K)} \right) = \frac{K}{40 + K} < 1 \quad (3)$$

*Eq. (3)* indicates that *large values of K lead to small steady-state error*; however, as seen below, they also lead to a *faster, less damped responses*.

The root locus diagram for the closed-loop system for  $K \geq 0$  and the Bode diagram for  $GH(s)$  are shown in *Fig. 2*. Note that as the value of  $K$  is *increased*, the closed-loop poles move straight up/down, indicating that the natural frequency is *increased*, and the damping ratio is *decreased*. Also, as the value of  $K$  is *increased*, the *phase (stability) margin is decreased*.

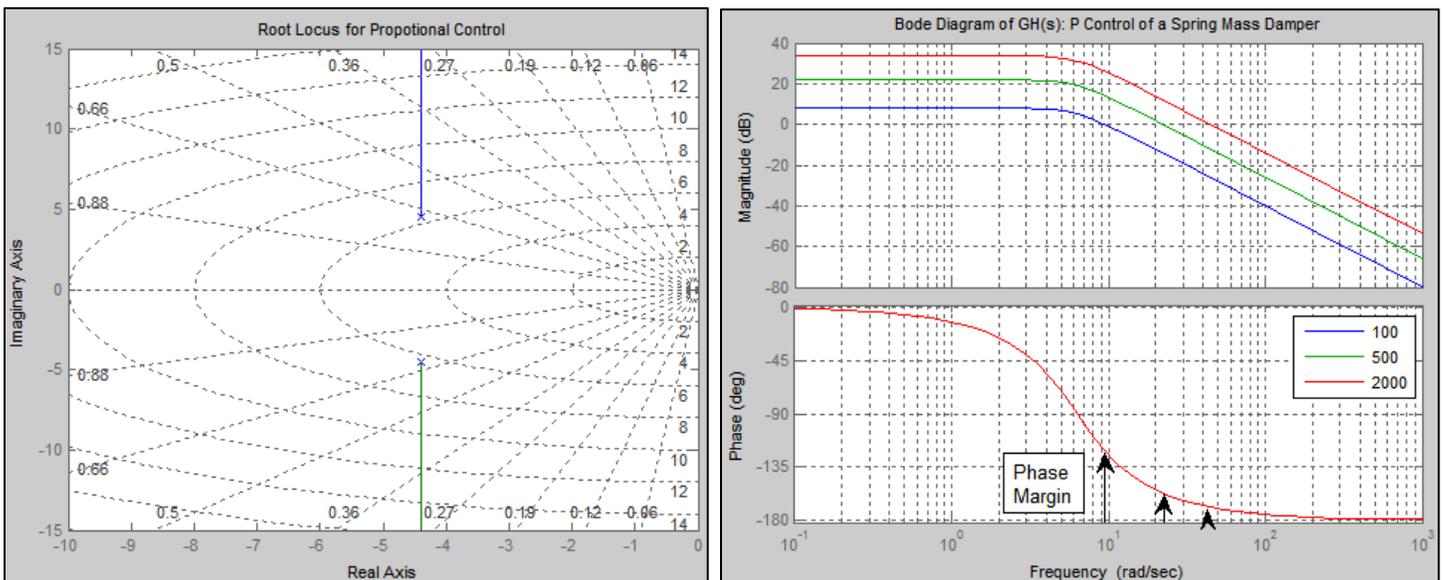


Figure 2. Root Locus Diagram and Bode Diagram for ( $GH(s)$ ) for Proportional Control

**Fig. 3** shows *step responses* and *Bode diagrams* of the closed-loop system for proportional gains  $K$  of 100, 500, and 2000. As the gain is *increased* the system time response is *faster* and *less damped*. The Bode diagram correspondingly shows *larger bandwidths* and *larger resonant magnitudes*. Clearly, it is *not possible* to achieve low steady-state error and good transient response using only proportional control. As the gain is increased, the response becomes faster, but it has a lower phase margin. To remove the steady-state error and have better response, integral and/or derivative terms must be included in the controller.

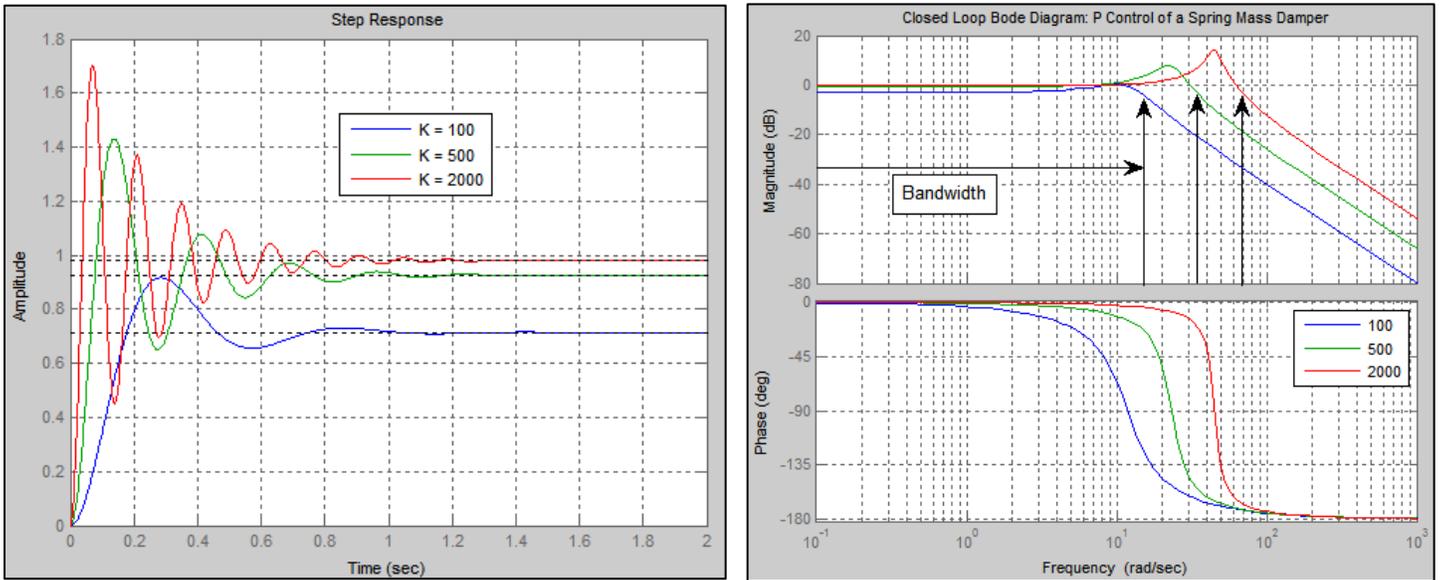


Figure 3. Closed Loop Step Response and Bode Diagrams for P Control

### Proportional-Integral (PI) Control

If *proportional-integral (PI) control* is used, the controller transfer function is

$$G_c(s) = K_p + \frac{K_I}{s} = \frac{K_p(s + a)}{s} \quad (4)$$

Here  $K_p$  and  $K_I$  represent the *proportional* and *integral* gains, and  $a = K_I/K_p$  is the ratio of the integral and proportional gains. The loop and closed-loop transfer functions for this system are

$$GH(s) = \frac{K_p(s + a)}{s(s^2 + 8.8s + 40)} \quad \frac{X}{X_d}(s) = \frac{K_p(s + a)}{s(s^2 + 8.8s + 40) + K_p(s + a)} \quad (5)$$

**Integral control** makes the system a **type-one** system, so the **steady-state error** due to a step input is **zero**. This can be verified by using the final value theorem to show that  $x_{ss} = 1$  when the input is a unit step function.

The root locus diagram for the closed-loop system (with  $a=3$ ) for  $K \geq 0$  and the Bode diagram for  $GH(s)$  are shown in **Fig. 4**. The root locus diagram also shows the locations of the closed-loop poles for a proportional gain  $K_p = 50$ . Note that the integral controller has **added a third, slower pole** to the system and has **moved the asymptotes** of the complex poles closer to the imaginary axis. For low gains, the system is **slow and stable** (first order dominant). As the gain is **increased**, the system becomes **faster** with a **decreasing phase margin**. The Bode diagram shows that the gain could be increased somewhat above  $K_p = 25$  without significantly decreasing the stability margin. However, further increases will decrease the phase margin.

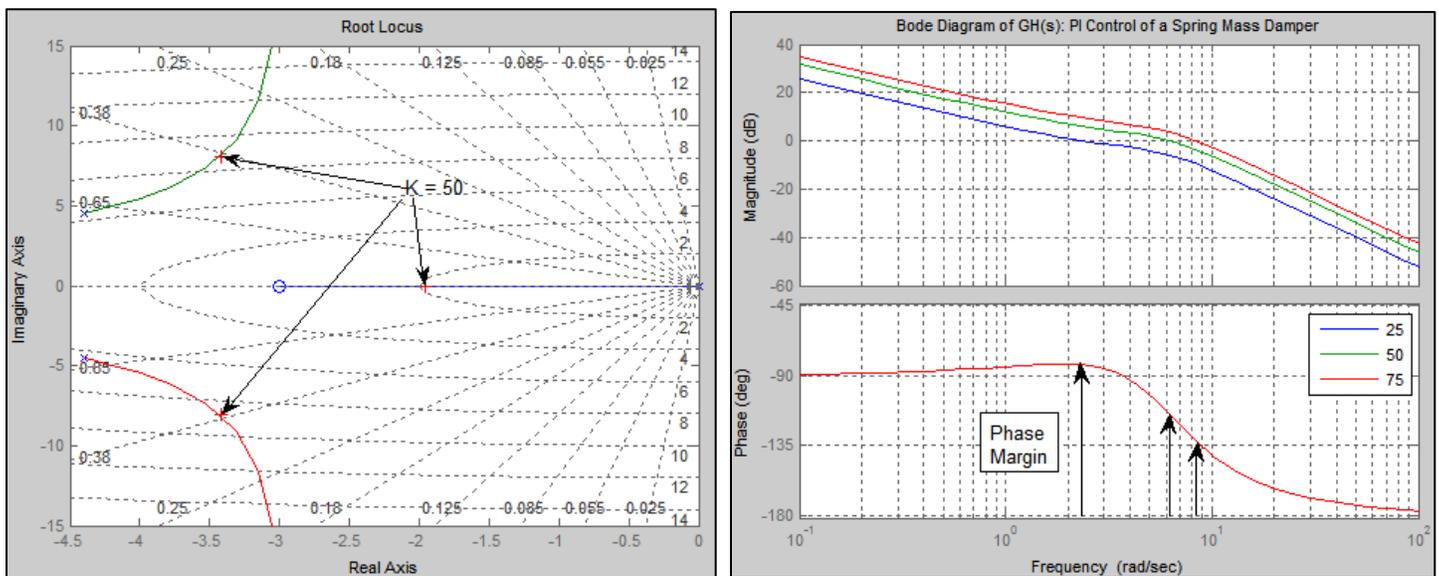


Figure 4. Root Locus Diagram and Bode Diagram for  $(GH(s))$  for PI Control ( $a=3$ )

**Fig. 5** shows step responses and Bode diagrams of the closed loop system for  $a=3$  and proportional gains of  $K_p = 25, 50,$  and  $75$ . Integral control has **removed the steady-state error** and **improved the transient response**, but it has also **increased the system settling time**. Settling times can be lowered by increasing the gain. This will **increase the system bandwidth**, but it will also decrease the stability margin.

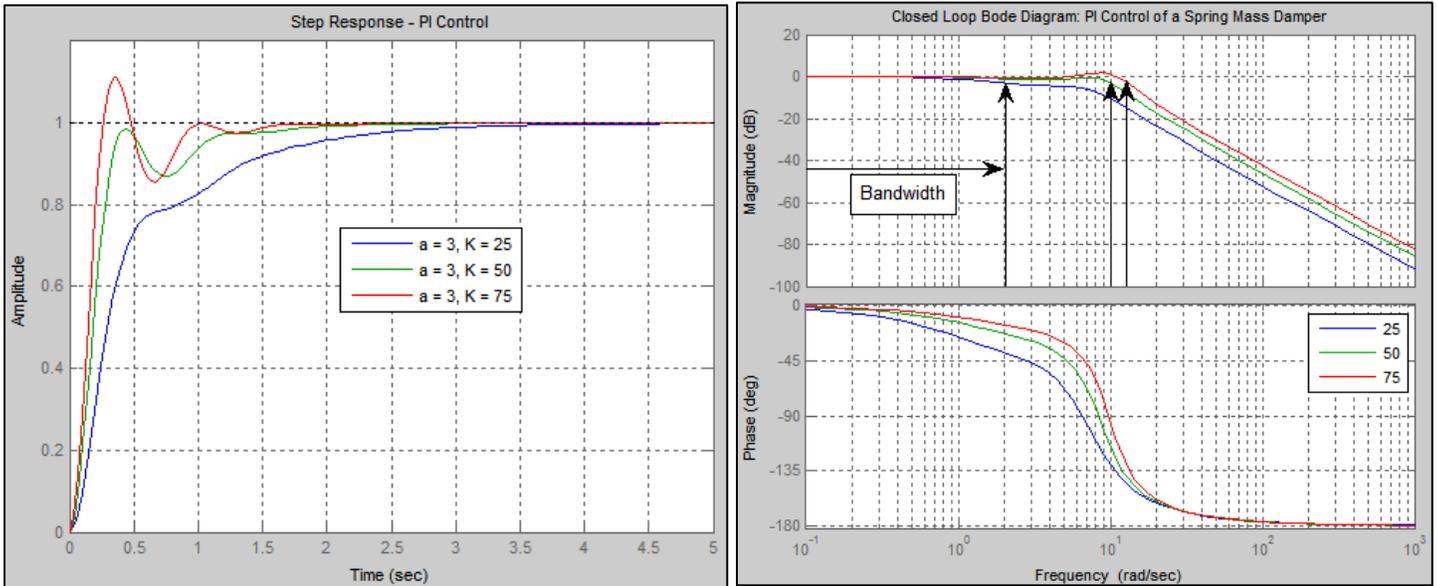


Figure 5. Closed Loop Step Response and Bode Diagrams for PI Control ( $a = 3$ )

### Proportional-Derivative (PD) Control

If *proportional-derivative (PD) control* is used, the controller transfer function is

$$G_c(s) = K_p + K_D s = K_D(s + a) \quad (6)$$

Here  $K_p$  and  $K_D$  represent the proportional and derivative gains, and  $a = K_p/K_D$  is the ratio of the proportional and derivative gains. The loop and closed-loop transfer functions for this system are

$$GH(s) = \frac{K_D(s + a)}{s^2 + 8.8s + 40} \quad \frac{X}{X_d}(s) = \frac{K_D(s + a)}{(s^2 + 8.8s + 40) + K_D(s + a)} \quad (7)$$

Without the integral control, this is again a *type-zero* system, and hence will have a *finite steady-state error* to a step input. Using the *final-value theorem* and the closed-loop transfer function,  $x_{ss}$  the final value of  $x(t)$  to a unit step command is

$$x_{ss} = \lim_{s \rightarrow 0} \left( s \cdot \frac{1}{s} \cdot \frac{K_D(s + a)}{(s^2 + 8.8s + 40) + K_D(s + a)} \right) = \frac{K_D a}{40 + K_D a} = \frac{K_p}{40 + K_p} < 1 \quad (8)$$

As with simple proportional control, the *larger the proportional gain*, the *smaller the steady-state error*.

The root locus diagram for the closed-loop system (with  $a = 10$ ) for  $K \geq 0$  and the Bode diagram for  $GH(s)$  are shown in *Fig. 6*. The root locus diagram also shows the locations of the

closed-loop poles for a derivative gain  $K_D \approx 25.6$ . As the gain is *increased* the system poles become *faster* and *more damped*. The Bode diagram indicates that the phase margin never drops below 90 degrees indicating a very stable system for any gain.

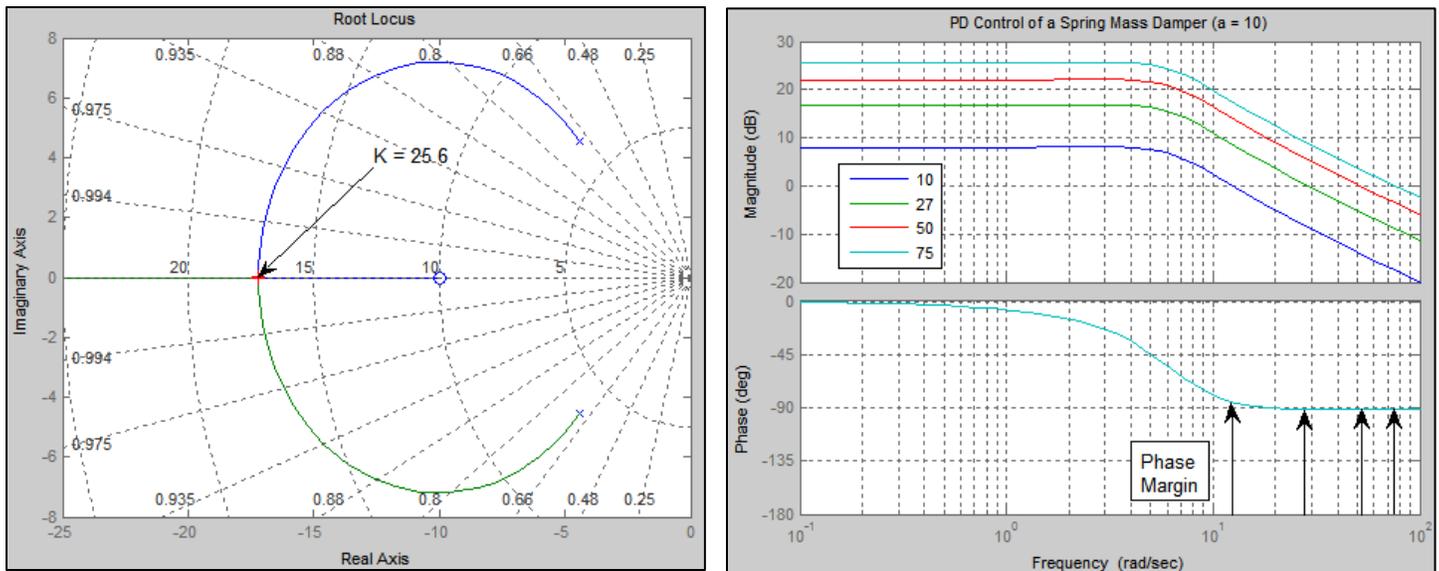


Figure 6. Root Locus Diagram and Bode Diagram for  $(GH(s))$  for PD Control ( $a = 10$ )

**Fig. 7** shows step responses and Bode diagrams of the closed loop system for  $a = 10$  and derivative gains of  $K_D = 10, 27, 50,$  and  $75$ . The PD controller has *decreased the system settling time* considerably; however, to control the steady-state error, the derivative gain  $K_D$  must be high. This will *decrease the response times and increase the bandwidth* of the system and may make it *susceptible to noise*.

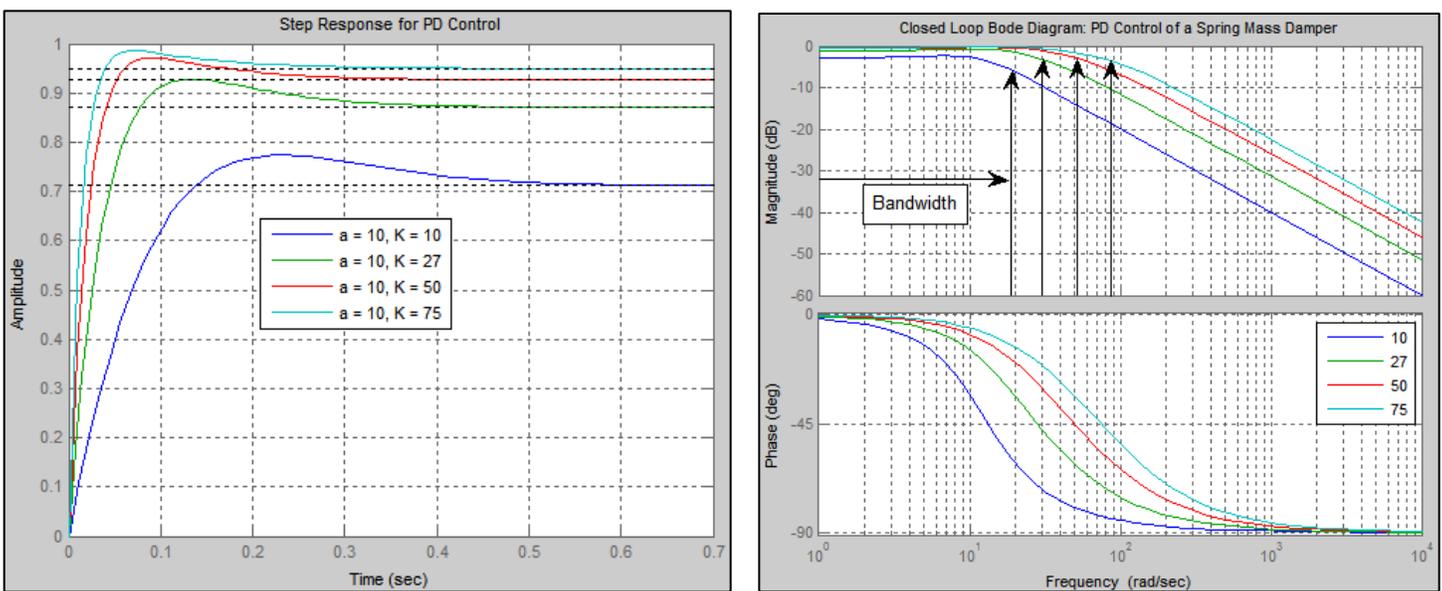


Figure 7. Closed Loop Step Response and Bode Diagrams for PD Control ( $a = 10$ )

## Proportional-Integral-Derivative Control

If *proportional-integral-derivative (PID) control* is used, the controller transfer function is

$$G_c(s) = K_p + \frac{K_I}{s} + K_D s = \frac{K_D(s^2 + as + b)}{s} \quad (9)$$

Here  $K_p$ ,  $K_I$ , and  $K_D$  represent the proportional, integral, and derivative gains,  $a = K_p/K_D$  is the ratio of the proportional and derivative gains, and  $b = K_I/K_D$  is the ratio of the integral and derivative gains. The loop and closed-loop transfer functions for this system are

$$GH(s) = \frac{K_D(s^2 + as + b)}{s(s^2 + 8.8s + 40)} \quad \frac{X}{X_d}(s) = \frac{K_D(s^2 + as + b)}{s(s^2 + 8.8s + 40) + K_D(s^2 + as + b)} \quad (10)$$

Again, with integral control, the system is *type-one* and has zero steady-state error for a step input.

The root locus diagram for the closed-loop system (with  $a = 15$  and  $b = 50$ ) for  $K \geq 0$  and the Bode diagram for  $GH(s)$  are shown in **Fig. 8**. The locations of the closed-loop poles for  $K_D \approx 15.8$  are also shown. As the gain is *increased*, the system becomes *faster without significant losses in the phase margin*.

**Fig. 9** shows step responses and Bode diagrams of the closed-loop system for  $a = 15$ ,  $b = 50$ , and derivative gains of  $K_D = 5, 10$ , and  $15$ . Using both integral and derivative control has *removed steady-state error* and *decreased system settling times* while maintaining a reasonable transient response.

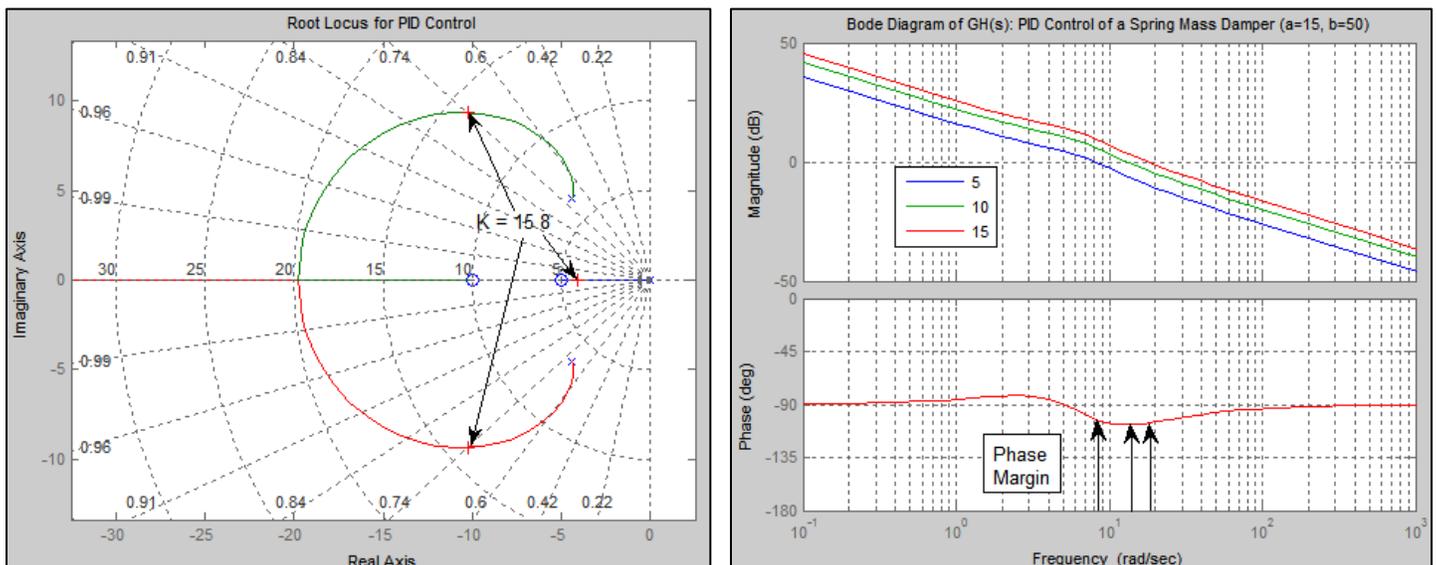


Figure 8. Root Locus Diagram and Bode Diagram for ( $GH(s)$ ) for PID Control ( $a = 15, b = 50$ )

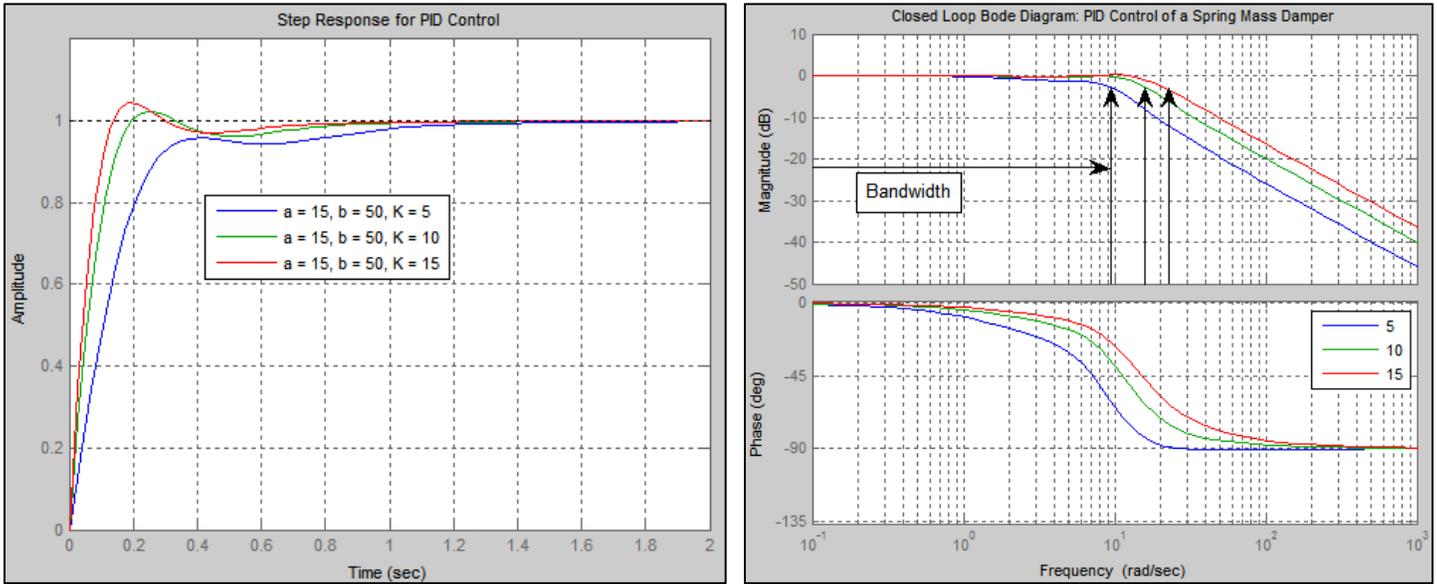


Figure 9. Closed Loop Step Response and Bode Diagrams for PID Control ( $a = 15, b = 50$ )