

Introductory Motion and Control

PID Position Control of a Spring-Mass-Damper: Analysis Summary

Fig. 1 shows a spring-mass-damper system with a force actuator for position control. The spring has stiffness k , the damper has coefficient c , the block has mass m , and the position of the mass is measured by the variable x . The *transfer function* of the SMD with the actuating force F_a as input and the position x as output is

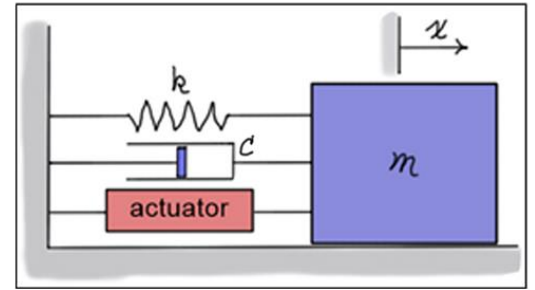
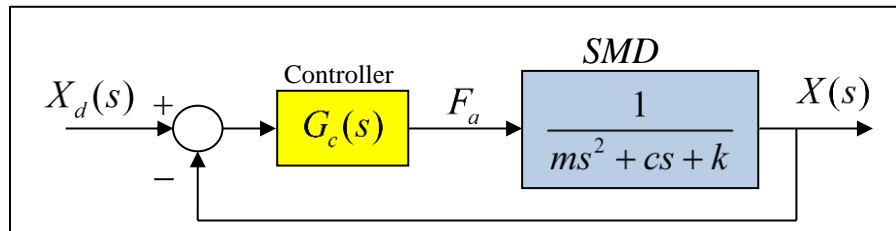


Fig. 1. Spring-Mass-Damper System with Force Actuator

$$\boxed{\frac{X}{F_a}(s) = \frac{1}{ms^2 + cs + k}} \quad (1)$$

Assuming ideal actuator and sensor responses, the closed-loop position control of the SMD can be described using the following block diagram. Here, X_d represents the *desired position*, and $G_c(s)$ represents the *transfer function* of the controller.



It is assumed here that the SMD parameters are: $m = 1$ slug, $c = 8.8$ (lb-s/ft), and $k = 40$ (lb/ft). This represents an under-damped, second-order plant with the following *natural frequency* and *damping ratio*.

$$\boxed{\begin{aligned} \omega_n &= \sqrt{40} = 6.325 \text{ (rad/s)} \approx 1 \text{ (Hz)} \\ \zeta &= \frac{8.8}{2\sqrt{40}} = 0.696 \approx 0.7 \end{aligned}}$$

Proportional Control

If *proportional control* is used, then $G_c(s) = K$, and the *loop transfer function* and *closed-loop transfer functions* are

$$\boxed{GH(s) = \frac{K}{s^2 + 8.8s + 40}} \quad \boxed{\frac{X}{X_d}(s) = \frac{K}{s^2 + 8.8s + (40 + K)}} \quad (2)$$

This is a *type-zero* system and will have a *finite steady-state error* for a step input. Using the final-value theorem and the closed-loop transfer function, x_{ss} the *final value* of $x(t)$ to a *unit step* command is

$$x_{ss} = \lim_{s \rightarrow 0} \left(s \cdot \frac{1}{s} \cdot \frac{K}{s^2 + 8.8s + (40 + K)} \right) = \frac{K}{40 + K} < 1 \quad (3)$$

Eq. (3) indicates that *large values* of K lead to *smaller steady-state errors*; however, as seen below, they also lead to a *faster, less damped responses*.

The root locus diagram for the closed-loop system for $K \geq 0$ and the Bode diagram for $GH(s)$ are shown in *Fig. 2*. Note that as the value of K is *increased*, the closed-loop poles move straight up/down, indicating the natural frequency is *increased* and the damping ratio is *decreased*. Also, as the value of K is *increased*, the *phase (stability) margin is decreased*.

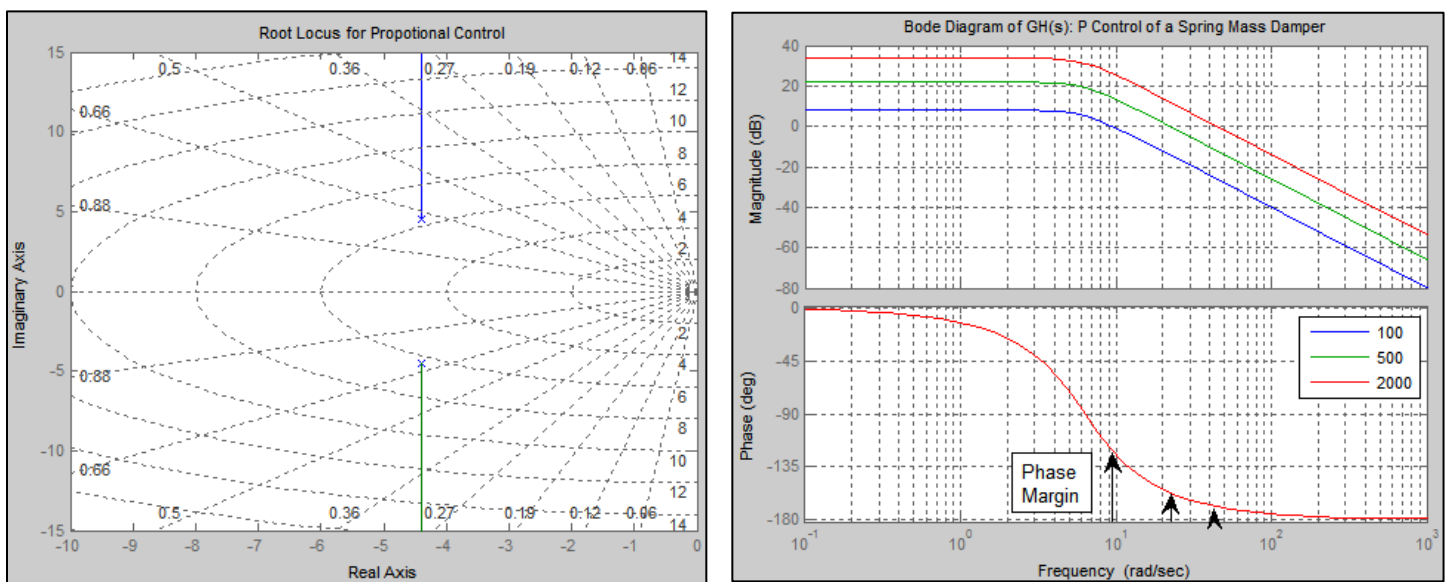


Fig. 2. Root Locus Diagram and Bode Diagram for $(GH(s))$ for Proportional Control

Fig. 3 shows step responses and Bode diagrams of the closed-loop system for proportional gains K of 100, 500, and 2000. As the gain is *increased* the system response is *faster* and *less damped*. The Bode diagram correspondingly shows *larger bandwidths* and *larger resonant magnitudes*. Clearly, it is not possible to achieve low steady-state error and good transient response using only proportional control. As the gain is increased, the response becomes *faster*, but it has a *lower phase margin*. To remove the steady-state error and have better response, integral and/or derivative terms must be included in the controller.

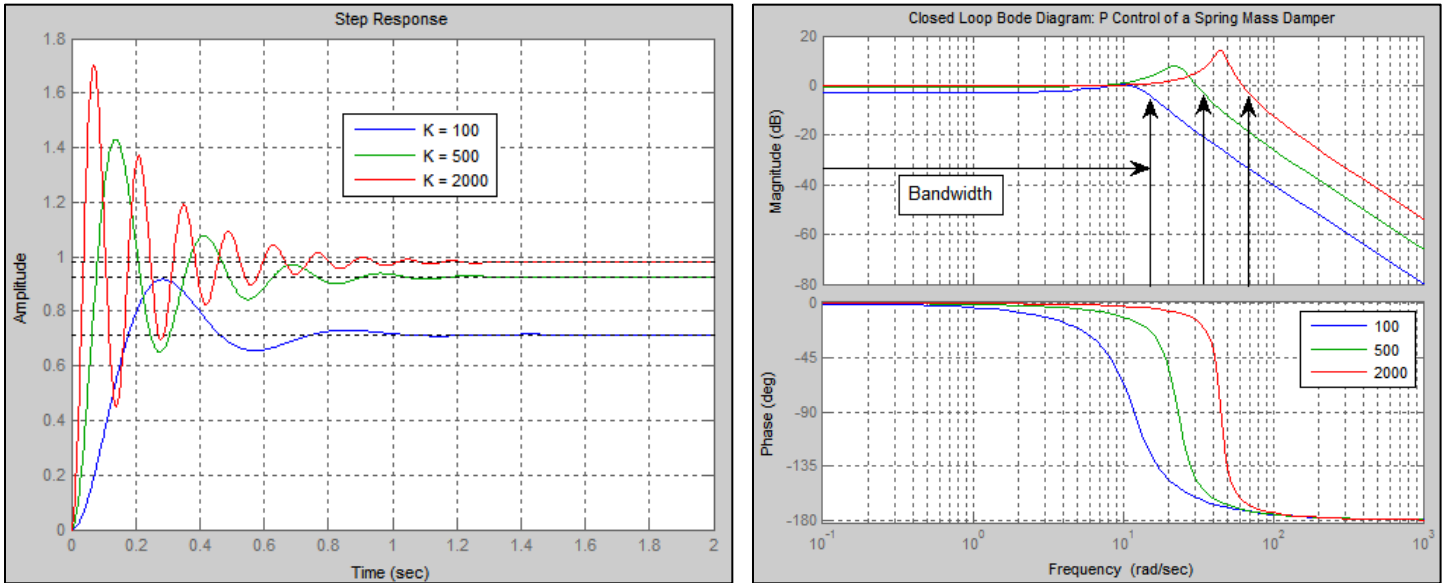


Fig. 3. Closed Loop Step Response and Bode Diagrams for P Control

Proportional-Integral (PI) Control

If *proportional-integral (PI) control* is used, then

$$G_c(s) = K_p + \frac{K_I}{s} = \frac{K_p(s + a)}{s} \quad (4)$$

The parameters K_p and K_I represent the *proportional* and *integral* gains, and the parameter $a = K_I/K_p$ is the *ratio* of the integral and proportional gains. The loop and closed-loop transfer functions for this system are

$$GH(s) = \frac{K_p(s + a)}{s(s^2 + 8.8s + 40)} \quad \frac{X}{X_d}(s) = \frac{K_p(s + a)}{s(s^2 + 8.8s + 40) + K_p(s + a)} \quad (5)$$

Integral control makes the system a *type-one* system, so the *steady-state error* due to a step input is *zero*. This can be verified using the final value theorem to show that $x_{ss} = 1$ for a unit step input.

The root locus diagram for the closed-loop system (with $a = 3$) for $K \geq 0$ and the Bode diagram for $GH(s)$ are shown in **Fig. 4**. The root locus diagram also shows the locations of the closed-loop poles for a proportional gain $K_p = 50$. Note the integral controller has *added a third, slower pole* to the system and has *moved the asymptotes* of the complex poles closer to the imaginary axis. For low gains, the system is *slow and stable* (first order dominant).

As the proportional gain is *increased*, the system becomes *faster* with a *decreasing phase margin*. The Bode diagram shows that the gain could be increased somewhat above $K_p = 25$ without significantly decreasing the stability margin. However, further increases will decrease the phase margin.

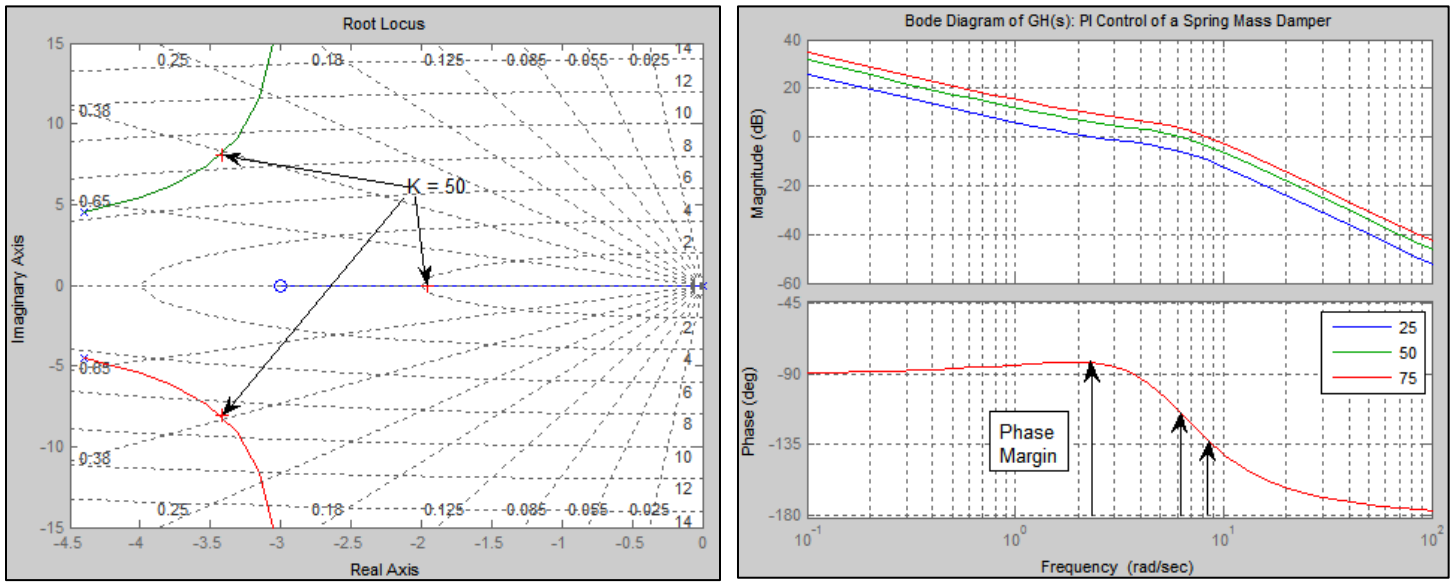


Fig. 4. Root Locus Diagram and Bode Diagram for ($GH(s)$) for PI Control ($a = 3$)

Fig. 5 shows step responses and Bode diagrams of the closed loop system for $a = 3$ and proportional gains of $K_p = 25, 50,$ and 75 . Integral control has *removed* the *steady-state step error* and *improved* the *transient response*, but it has also *increased* the *system's settling time*. Settling times can be lowered by *increasing* the gain. This will *increase* the *system bandwidth*, but it will also *decrease* the *stability margin*.

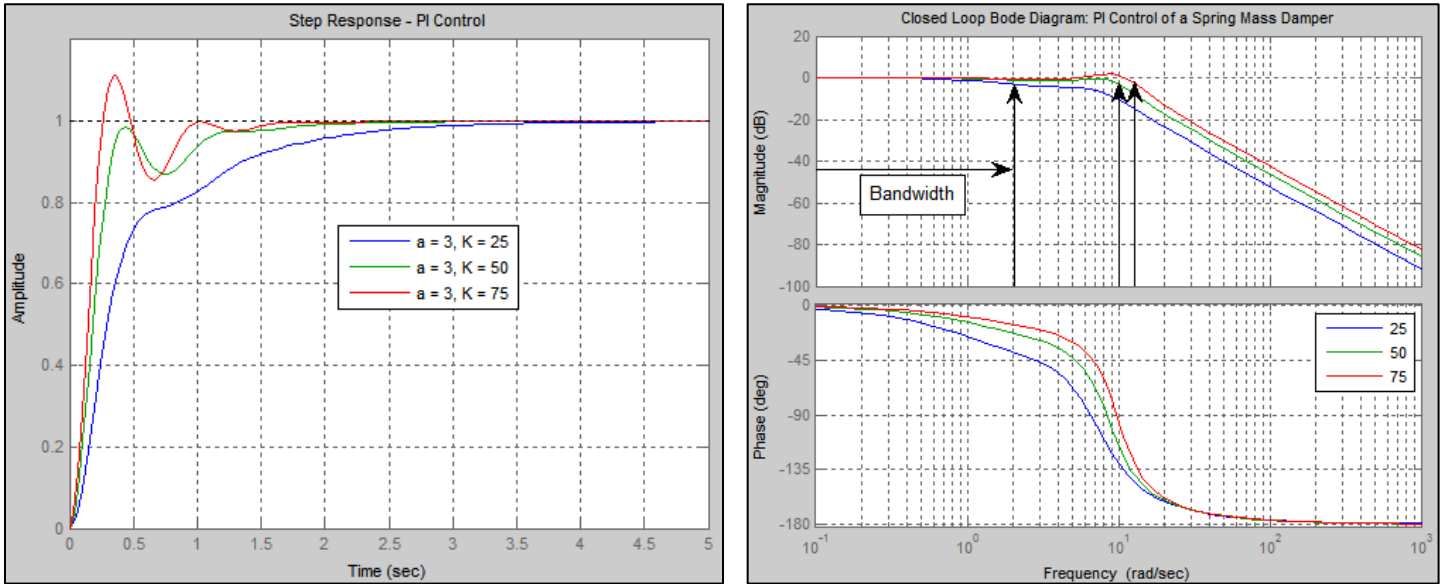


Fig. 5. Closed Loop Step Response and Bode Diagrams for PI Control ($a = 3$)

Proportional-Derivative (PD) Control

If *proportional-derivative (PD) control* is used, then

$$\boxed{G_c(s) = K_p + K_D s = K_D(s + a)} \quad (6)$$

The parameters K_p and K_D represent the *proportional* and *derivative* gains, and the parameter $a = K_p/K_D$ is the *ratio* of the proportional and derivative gains. The loop and closed-loop transfer functions for this system are

$$\boxed{GH(s) = \frac{K_D(s + a)}{s^2 + 8.8s + 40}} \quad \boxed{\frac{X}{X_d}(s) = \frac{K_D(s + a)}{(s^2 + 8.8s + 40) + K_D(s + a)}} \quad (7)$$

Without the integral control, this is again a *type-zero* system, and hence will have a *finite steady-state error* to a unit step input. Using the final-value theorem and the closed-loop transfer function, x_{ss} the final value of $x(t)$ to a unit step command is

$$x_{ss} = \lim_{s \rightarrow 0} \left(s \cdot \frac{1}{s} \cdot \frac{K_D(s + a)}{(s^2 + 8.8s + 40) + K_D(s + a)} \right) = \frac{K_D a}{40 + K_D a} = \frac{K_p}{40 + K_p} < 1 \quad (8)$$

As with simple proportional control, the *larger the proportional gain*, the *smaller the steady-state error*.

The root locus diagram for the closed-loop system (with $a = 10$) for $K \geq 0$ and the Bode diagram for $GH(s)$ are shown in **Fig. 6**. The root locus diagram also shows the locations of the

closed-loop poles for a derivative gain $K_D \approx 25.6$. As the gain is *increased* the system's poles become *faster* and *more damped*. The Bode diagram indicates that the phase margin never drops below 90 degrees indicating a *very stable* closed-loop system for any gain.

Fig. 7 shows step responses and Bode diagrams of the closed loop system for $a = 10$ and derivative gains of $K_D = 10, 27, 50,$ and 75 . The PD controller has *decreased* the *system's settling time* considerably. However, to control the steady-state error, the derivative gain K_D must be high. This will *decrease* the *response times* and *increase* the *bandwidth* of the system and may make it *susceptible to noise*.

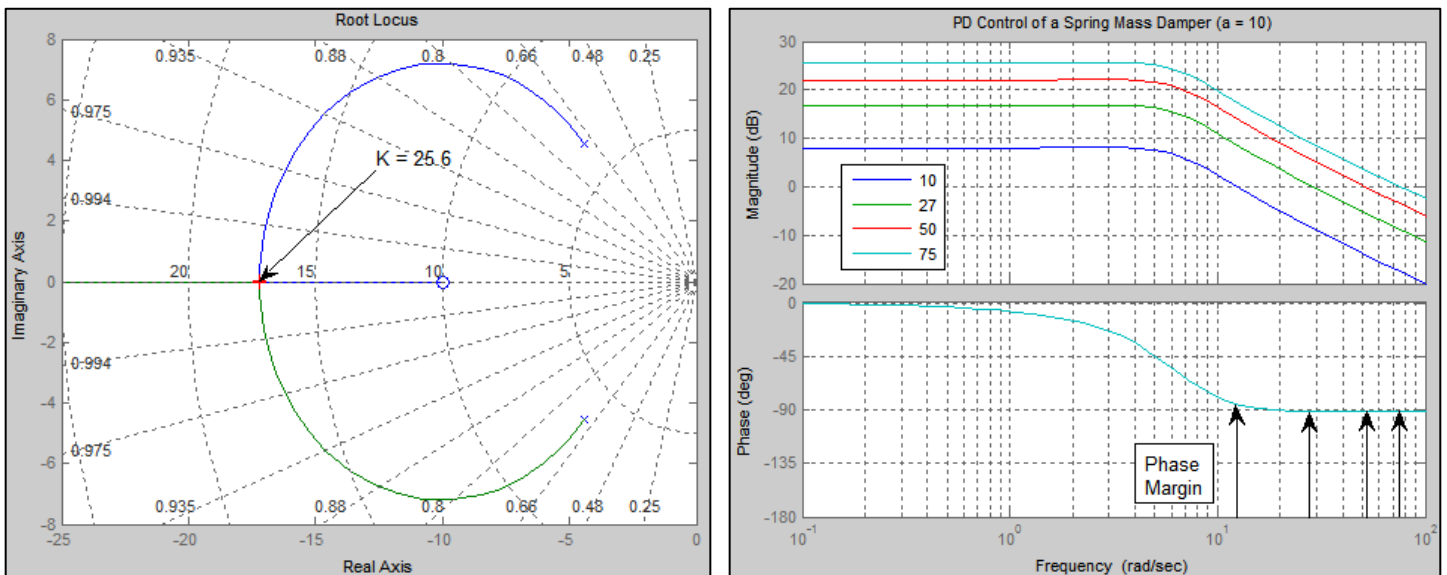


Fig. 6. Root Locus Diagram and Bode Diagram for $(GH(s))$ for PD Control ($a = 10$)

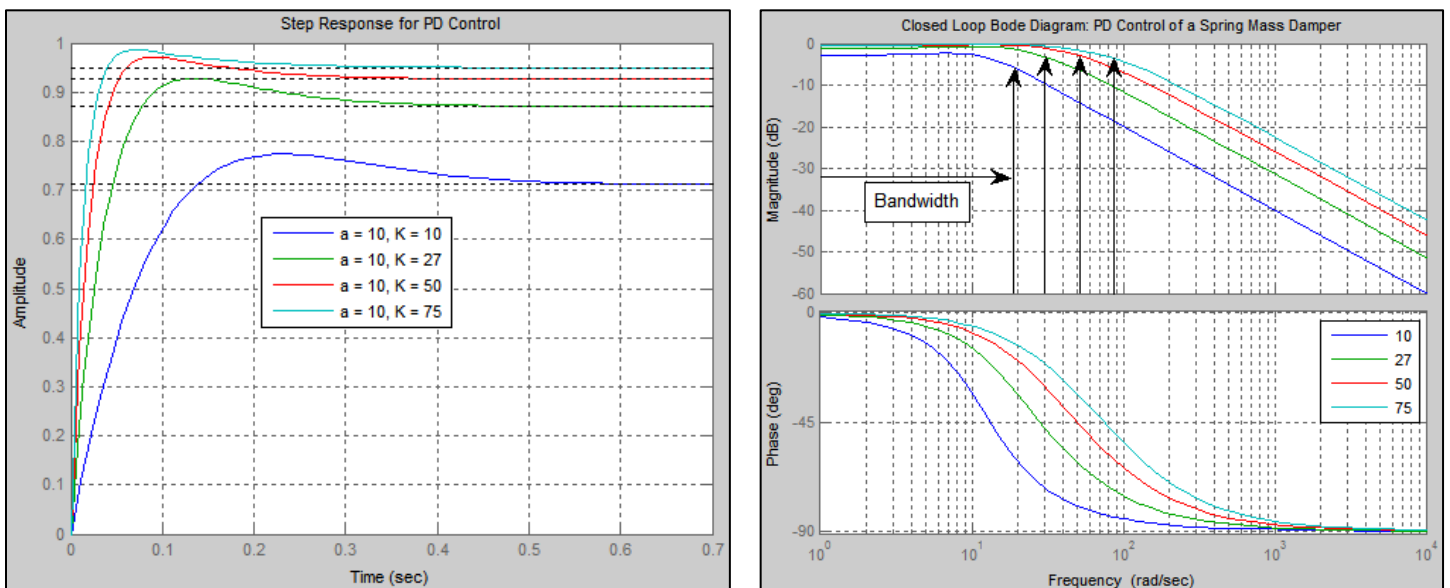


Fig. 7. Closed Loop Step Response and Bode Diagrams for PD Control ($a = 10$)

Proportional-Integral-Derivative Control

If *proportional-integral-derivative (PID) control* is used, then

$$G_c(s) = K_p + \frac{K_I}{s} + K_D s = \frac{K_D(s^2 + as + b)}{s} \quad (9)$$

The parameters K_p , K_I , and K_D represent the *proportional*, *integral*, and *derivative* gains, the parameter $a = K_p/K_D$ is the *ratio* of the proportional and derivative gains, and the parameter $b = K_I/K_D$ is the *ratio* of the integral and derivative gains.

The loop and closed-loop transfer functions for this system are

$$GH(s) = \frac{K_D(s^2 + as + b)}{s(s^2 + 8.8s + 40)} \quad \frac{X}{X_d}(s) = \frac{K_D(s^2 + as + b)}{s(s^2 + 8.8s + 40) + K_D(s^2 + as + b)} \quad (10)$$

Integral control makes the system *type-one*, so it has *zero* steady-state error for a step input.

The root locus diagram for the closed-loop system (with $a = 15$ and $b = 50$) for $K \geq 0$ and the Bode diagram for $GH(s)$ are shown in **Fig. 8**. The locations of the closed-loop poles for $K_D \approx 15.8$ are also shown. As the gain is *increased*, the system becomes *faster without significant losses in the phase margin*.

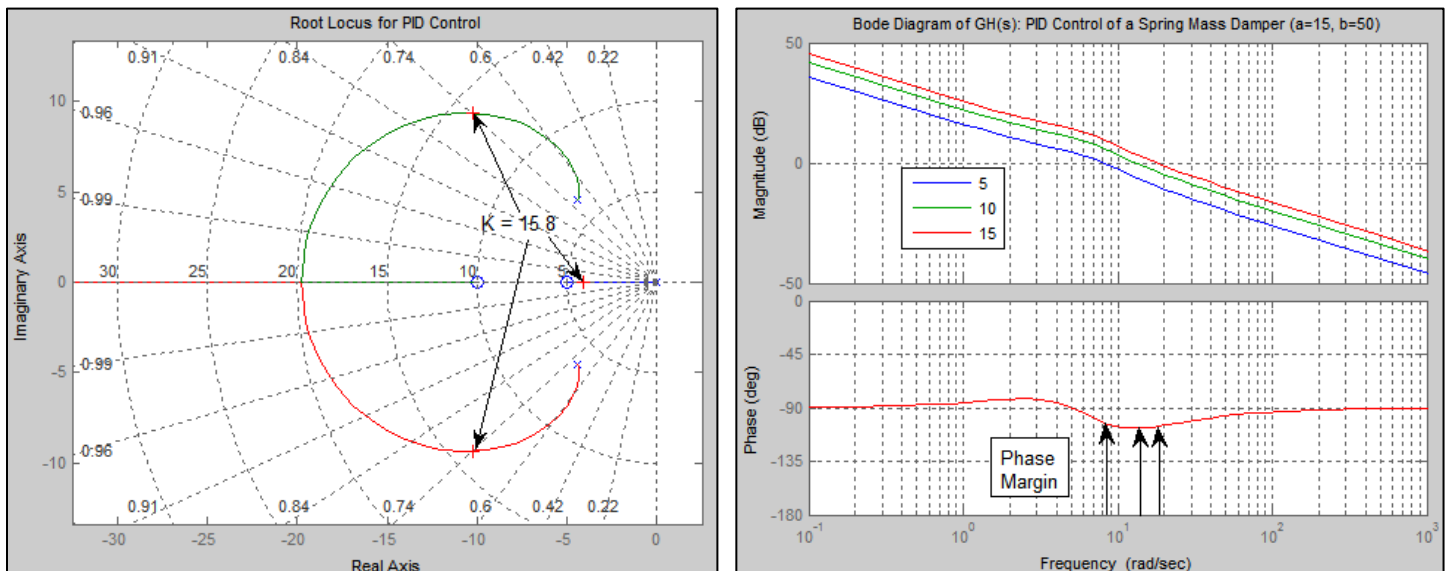


Fig. 8. Root Locus Diagram and Bode Diagram for ($GH(s)$) for PID Control ($a = 15, b = 50$)

Fig. 9 shows step responses and Bode diagrams of the closed-loop system for $a = 15$, $b = 50$, and derivative gains of $K_D = 5, 10$, and 15 . Using both integral and derivative control has *removed*

steady-state error and *decreased system settling times* while maintaining a reasonable transient response.

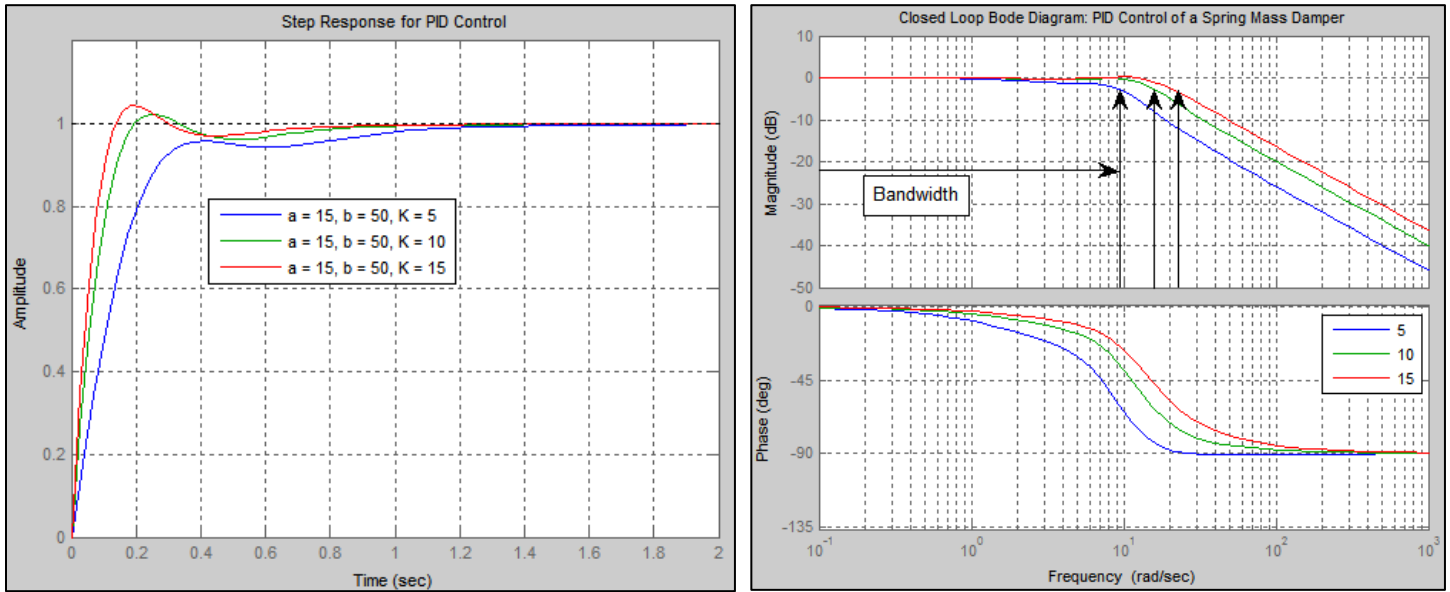


Fig. 9. Closed Loop Step Response and Bode Diagrams for PID Control ($a = 15, b = 50$)